
Rotordynamics Analysis Overview

Featuring Analysis Capability of RAPPID™

Prepared by

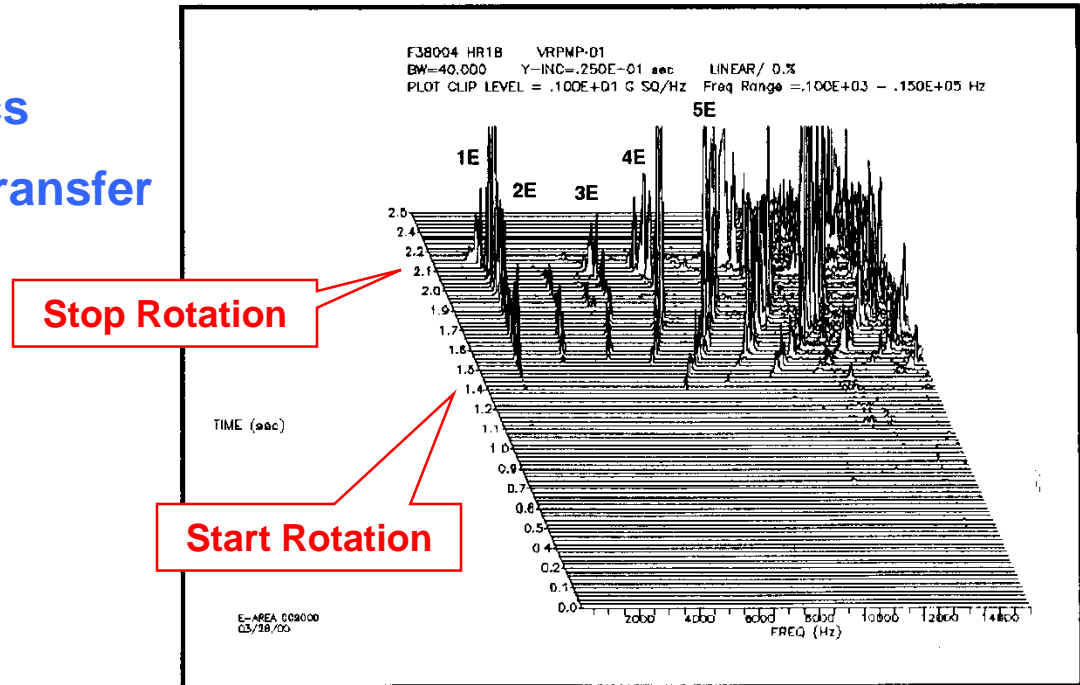
Rotordynamics-Seal Research

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Vibration of Rotating Systems

- **Vibration Signature can be Quite Complex in Nature**
- **Vibration Modeling is Frequently used to Aid Design & Development**
- **Vibration Model**
 - Structural Characteristics
 - Structure-to-Structure Transfer Functions
 - Forcing Functions
- **Vibration Signature**
 - Frequency
 - Magnitude
 - Phase



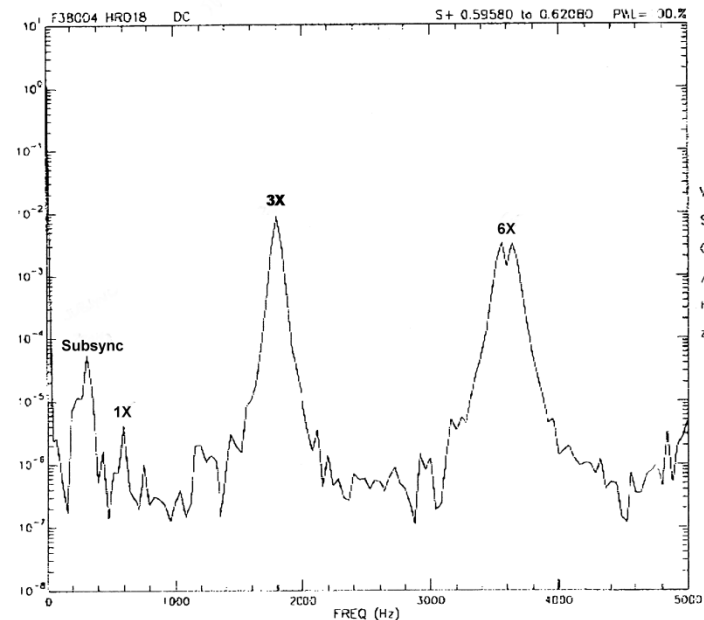
Vibration Signature Contributors

- **Major Peak Sources**

- Structural Harmonics
- Impacts/Rubs
- Misalignment
- Rotor Bow
- Unbalance
 - Mechanical
 - Hydraulic
- Trapped Fluids
- Unstable Rotor Whirl
- Large Scale Flow Effects
 - Periodic
 - Unsteady
 - Unstable

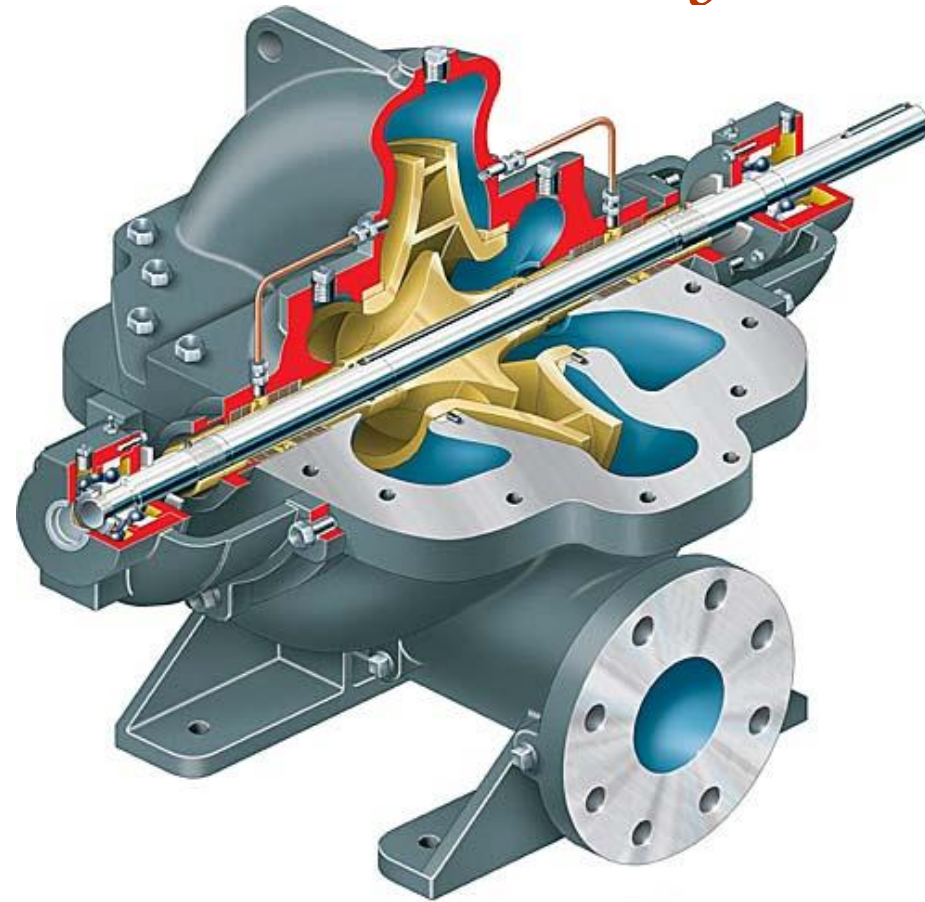
- **Noise Floor Sources**

- Acoustics
- Light Rubs
- Small Scale Flows Effects
 - Turbulent eddies
 - Cavitation bubbles
 - Localized reversals



Typical Commercial Machinery

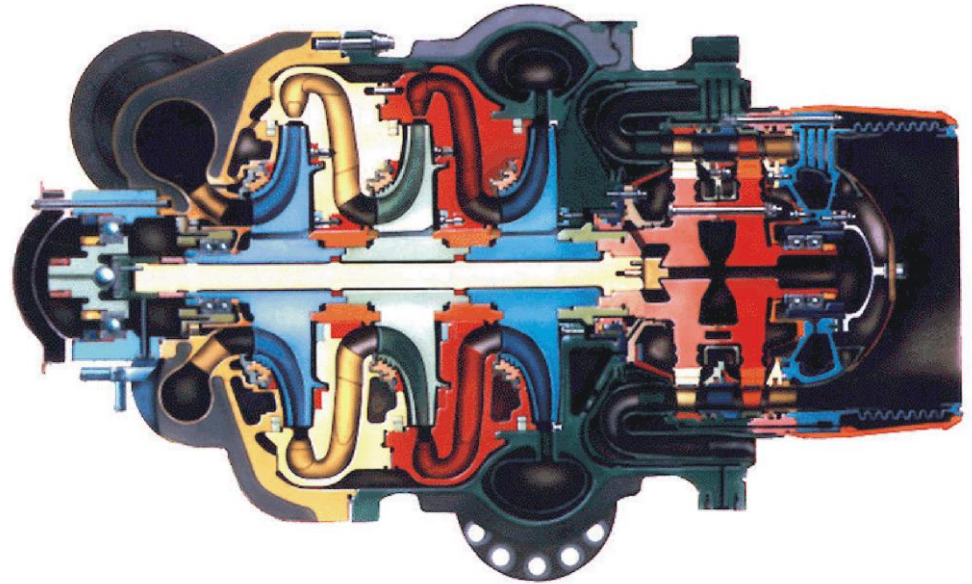
- **Stationary**
 - Few Weight/Volume Restrictions
 - Low speed
 - Rigid housing
 - Sub-critical rotors
 - Compartmentalized designs
 - Oil Lubrication Systems
 - Incompressible lubricant
 - Stiffness, damping, mass coef.
 - Viscous lubricant
 - High rotor damping
- **Low Energy Density**
 - Pump: 0.5 Hp/lb
 - Gas Turbine: 3 Hp/lb
- **Continuous Operation**
 - Steady State
 - Thermal Equilibrium
 - Constant Power Levels



Minimal Unsteady Effects

Typical Rocket Engine Turbopump

- **Mobile**
 - Extreme Weight/Volume Restrictions
 - High speed
 - Flexible rotors & housings
 - Highly integrated designs
 - Process Lubricated
 - Compressible lubricants
 - Transfer functions
 - Low viscosity lubricants
 - Lightly damped rotors
 - Varying Ambient Conditions
- **High Energy Density (~ 100 Hp/lb)**
 - Use of Cryogenics
 - Wide pressure/temperature ranges
 - Steep Ramp Rates
- **Short Run Durations**
 - Power Level Changes
 - Steady state never achieved



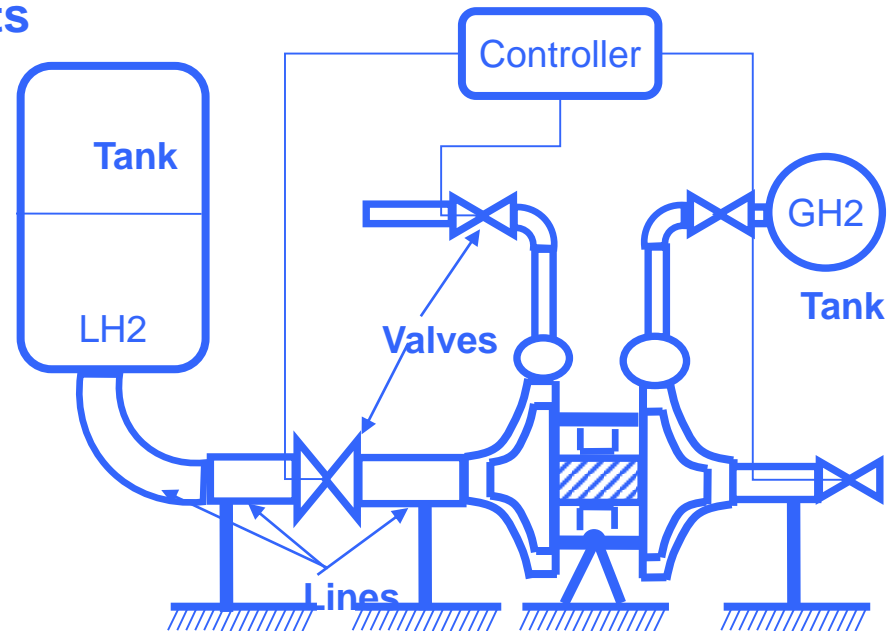
Major Unsteady Effects

Rotordynamics Model

Elements of Vibration Model	Sub Categories
Structural Characteristics	Rotating Assemblies
	Stationary Assemblies
Structure-to-Structure Transfer Functions	Fluidic Interfaces
	Mechanical Interfaces
	Hybrid Interfaces
Forcing Functions	Flow Related
	Mechanical Related
	Electrical/Magnetic Related
	Controls Related
	Rotor/Stator Interactions
Operating Geometry Changes	Distortions
	Relative Displacements

Structural Characteristics

- **Structural Characteristics Purpose**
 - Establish Structural Compliance and Resonance Frequencies
- **Required for Rotating and Stationary Assemblies**
 - Includes Facility/Engine Effects
- **Critical Phenomena**
 - Mass
 - Inertia
 - Load Path
 - Material Properties
 - Temperature dependent



Accurate structural characterization is critical for establishing natural frequency locations

Transfer Functions

- **Transfer Function Purpose**
 - Translate the Motion of One Structure into Forces on Another Structure
 - Motion measured in displacements, velocities, accelerations

$$\begin{Bmatrix} F_x \\ F_y \end{Bmatrix} = \begin{bmatrix} H_{xx}(\omega) & H_{xy}(\omega) \\ H_{yx}(\omega) & H_{yy}(\omega) \end{bmatrix} \begin{Bmatrix} X \\ Y \end{Bmatrix}$$

- **Required for all Rotating and Stationary Structure Interfaces**
 - Typical Interfaces Include
 - Bearings, seals, dampers
 - Pump, turbine, inducer wheels
 - Splines/couplings
 - Pump out vanes
 - Rub surfaces

Accurate transfer functions characterization is critical for establishing orbit stability and natural frequency locations

Transfer Functions

- **General Form of Transfer Function (H matrix)**
 - H Matrix Elements are Complex
 - H Matrix Elements May Vary Non-Linearly with Frequency

$$\begin{Bmatrix} F_x \\ F_y \end{Bmatrix} = \begin{bmatrix} H_{xx}(\omega) & H_{xy}(\omega) \\ H_{yx}(\omega) & H_{yy}(\omega) \end{bmatrix} \begin{Bmatrix} X \\ Y \end{Bmatrix}$$

– **Assuming** an Interface Adheres to the Linearized Model Leads To:

$$H_{xx}(\omega) = K_{xx} + iC_{xx}\omega - M_{xx}\omega^2$$

$$H_{xy}(\omega) = K_{xy} + iC_{xy}\omega - M_{xy}\omega^2$$

$$H_{yx}(\omega) = K_{yx} + iC_{yx}\omega - M_{yx}\omega^2$$

$$H_{yy}(\omega) = K_{yy} + iC_{yy}\omega - M_{yy}\omega^2$$

Forcing Functions

- **Forcing Function Purpose**
 - Excite the Rotating and Stationary Assemblies
 - Defined by magnitude, frequency, and phase
- **Required for Excitation Forces Acting on the Rotating and Stationary Structures**
 - Typical Excitation Forces Include
 - Rotor unbalance
 - Impacts/rubs
 - Misalignment, shaft bow, loose press fits
 - Hydraulic unbalance
 - Trapped fluid in a rotating structure
 - Steady and unsteady flow fluctuations
 - Valve induced
 - Controller induced
 - Controller imperfections
 - Rotor/stator interactions
 - Jet, vane pass, vortex shedding

Rotordynamic Analysis

- Available Analysis Types
 - Eigenvalue
 - Free-Free
 - Undamped Critical Speed
 - Damped Eigenvalue (Stability)
 - Forced Response (Linear)
 - Steady State
 - Forced Response (Non-Linear)
 - Transient (not covered in this information package)

Free-Free Analysis

- **Required Information**
 - Structural Model
- **Analysis Assumptions**
 - No Rotation
 - No Interconnection Forces
 - No Forcing Functions
- **Analysis Results**
 - Natural Frequencies
 - Mode Shapes (planar)
- **Why Perform Free-Free Analysis?**
 - Verify Structural Model by Comparing to Rap Test Data

Sample Rotor Model

ROTOR 1 STRUCTURE SCHEMATIC

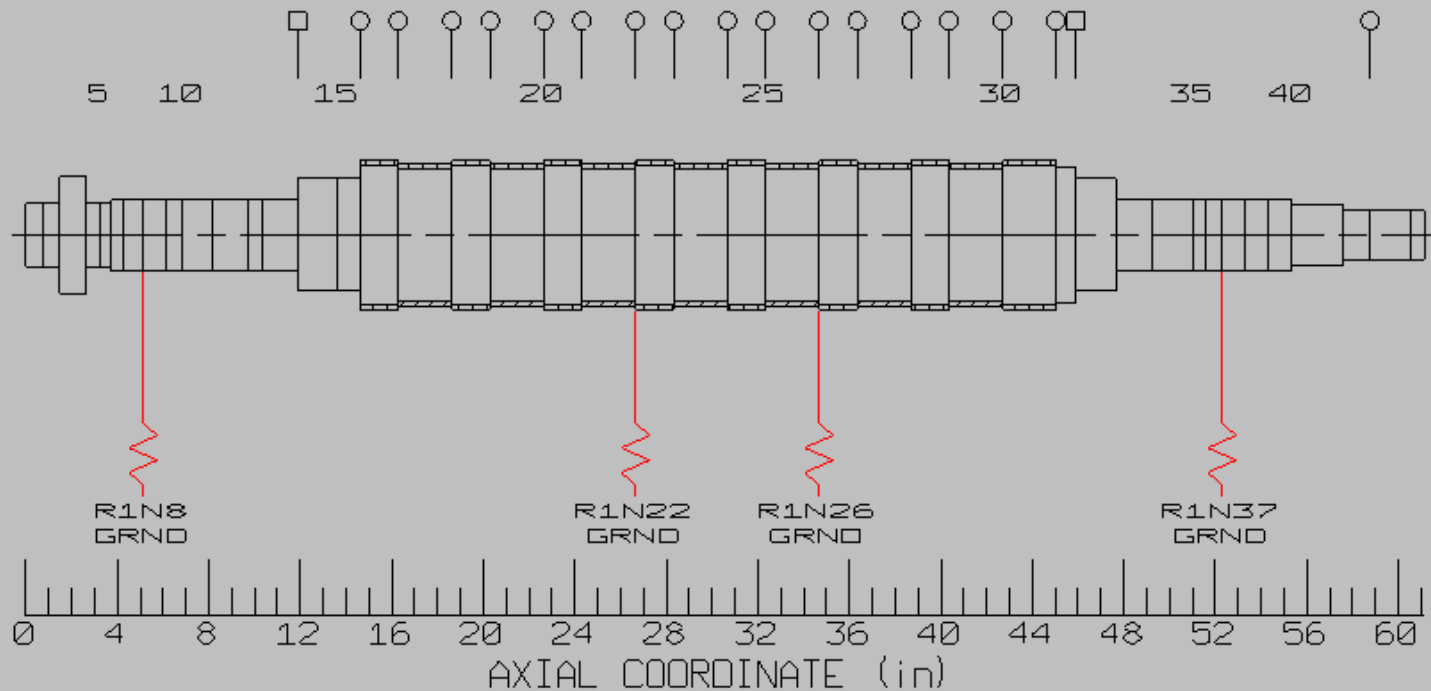
Example Multi-Stage Compressor

Eigenvalue Analysis as a Function of Rotor Speed

Entity Mass=694.086 lbm Entity Length=61.140 in C.G.=29.923 in

	DENSITY	E	G
	(lbm/in ³)	(psi)	(psi)
Material 1:	0.283	3.00e+007	1.20e+007
Material 2:	0.000	3.00e+007	1.20e+007

- Added Mass & Inertia
- Added Mass only
- △ Added Inertia only
- ✕ Excitation Applied



RAPPID (TM) 3.0 Wed Feb 27 16:35:43 2008

1st Free-Free Bending Mode

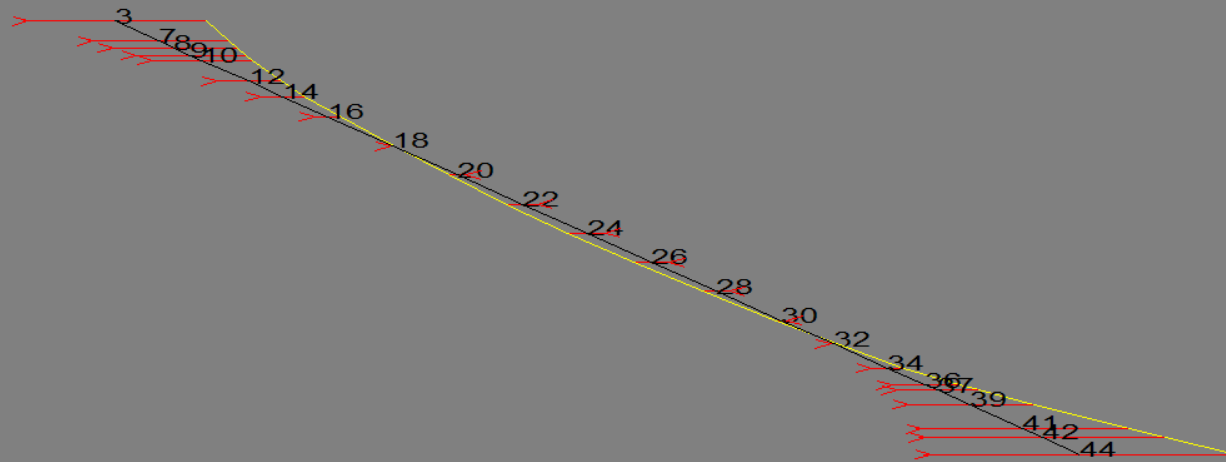
ROTORDYNAMIC MODE SHAPE PLOT

Manual Example Case

Multi-Stage Compressor

ANALYSIS POINT: ROTOR SPEED = 0 rpm

NAT FREQUENCY = 17946 cpm



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Mode Shape Plots (4) Plot # 1

2nd Free-Free Bending Mode

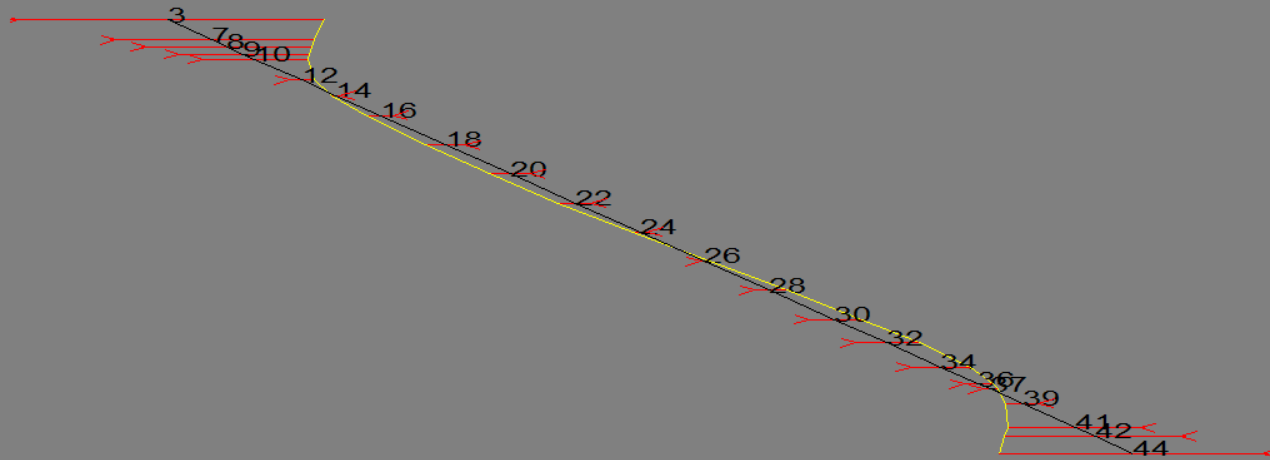
ROTORDYNAMIC MODE SHAPE PLOT

Manual Example Case

Multi-Stage Compressor

ANALYSIS POINT: ROTOR SPEED = 0 rpm

NAT FREQUENCY = 27886 cpm



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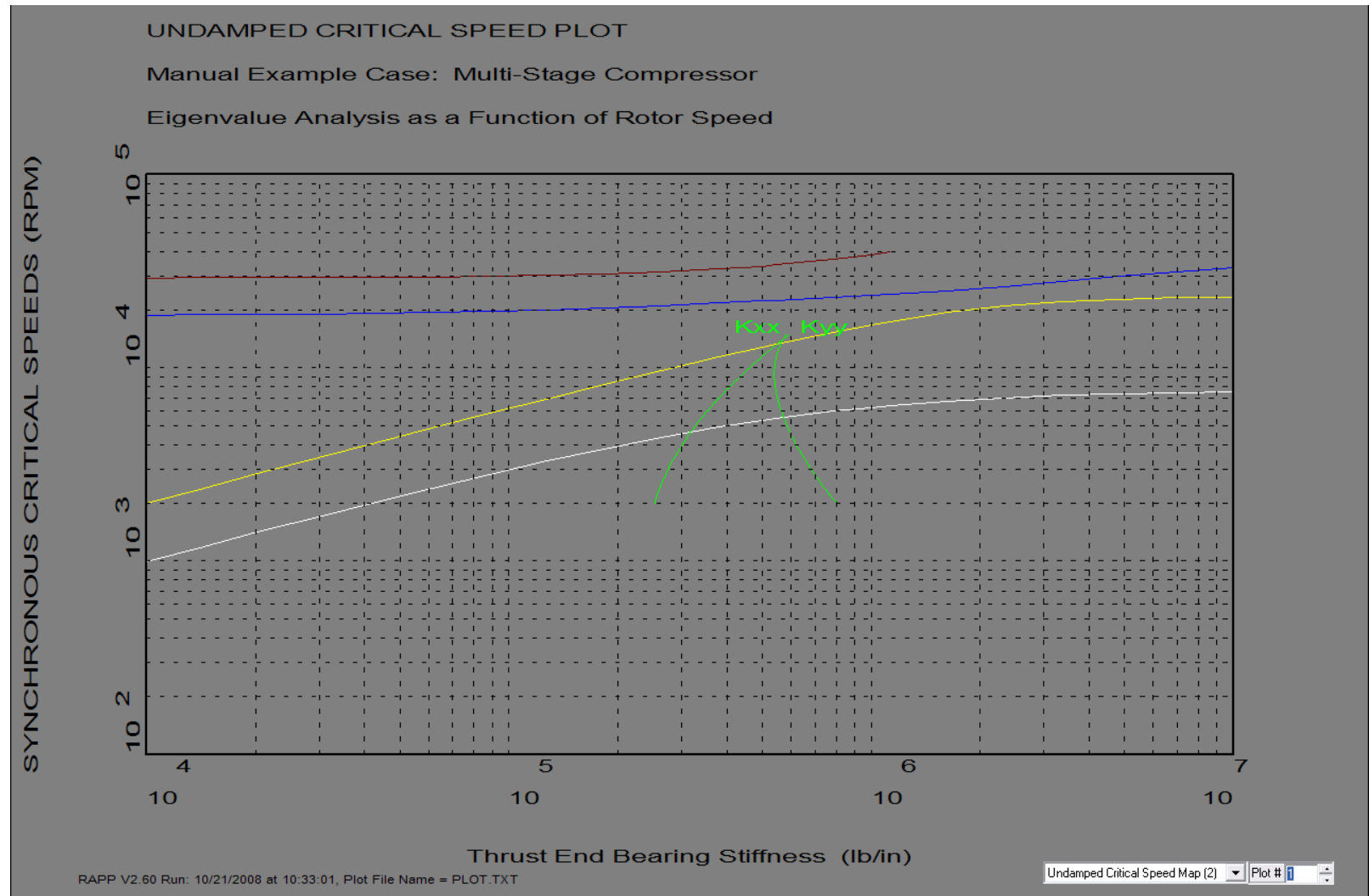
Mode Shape Plots (4)

Plot # 3

Undamped Critical Speed Analysis

- **Required Information**
 - Structural Model
 - Range of Bearing Stiffness
- **Analysis Assumptions**
 - No Damping
 - No Cross-Coupling
 - Symmetric Rotor Supports
 - Natural Frequency Coincides with Running Speed
- **Analysis Results**
 - Synchronous Critical Speed as a Function of Direct Stiffness
 - Mode Shapes (circular)
- **Why Perform Undamped Critical Speed Analysis?**
 - If Precise Transfer Functions are not Available

Undamped Critical Speed Map



1st Synchronous Critical Speed

ROTORDYNAMIC MODE SHAPE PLOT

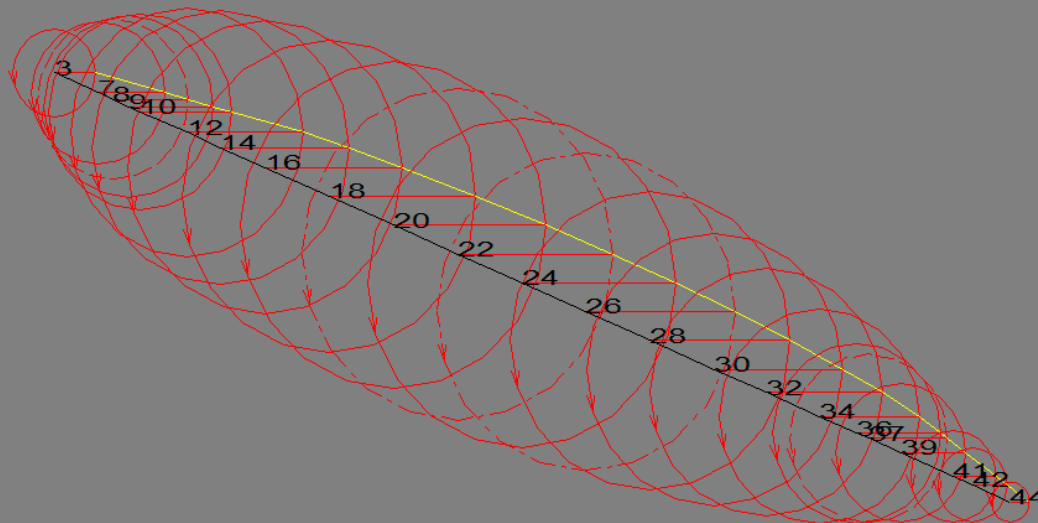
Manual Example Case: Multi-Stage Compressor

Eigenvalue Analysis as a Function of Rotor Speed

ANALYSIS POINT: Thrust End Bearing Stiffness = 500000 (lb/in)

CRITICAL SPEED = 5347 rpm

STATION 22 ORBIT FORWARD PRECESSION (FORWARD=RED, BACKWARD=BLUE)



RAPP V2.60 Run: 10/21/2008 at 10:33:01, Plot File Name = PLOT.TXT

Mode Shape Plots (8) Plot # 1

2nd Synchronous Critical Speed

ROTOR DYNAMIC MODE SHAPE PLOT

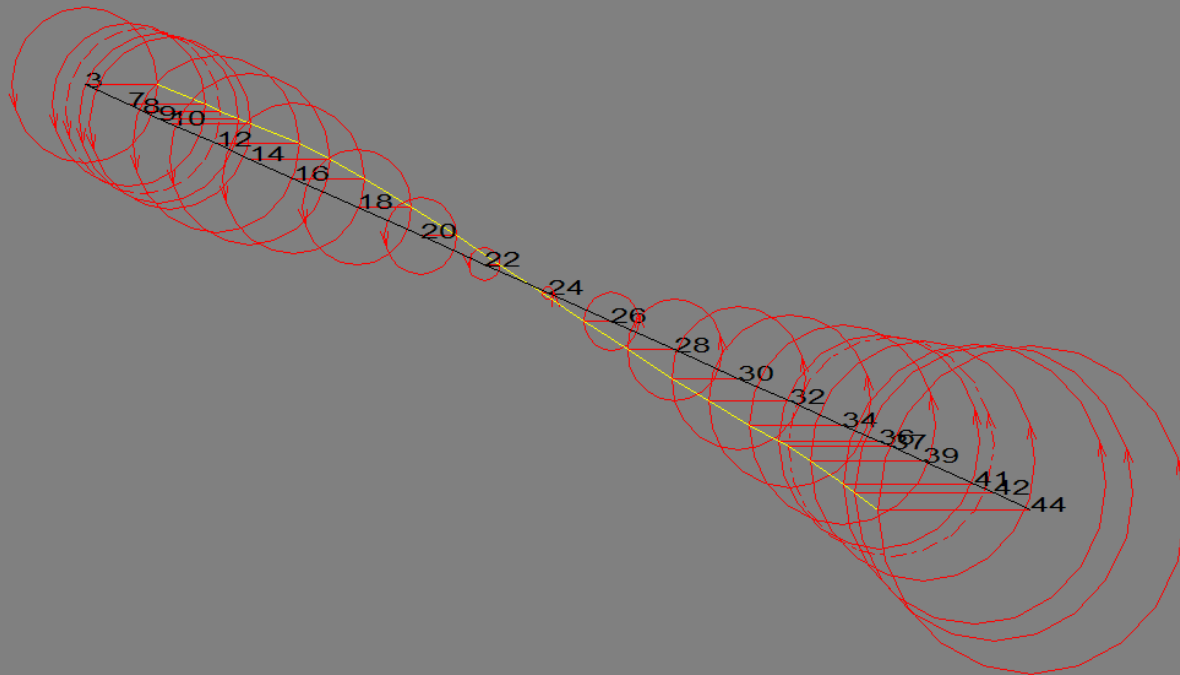
Manual Example Case: Multi-Stage Compressor

Eigenvalue Analysis as a Function of Rotor Speed

ANALYSIS POINT: Thrust End Bearing Stiffness = 500000 (lb/in)

CRITICAL SPEED = 12798 rpm

STATION 44 ORBIT FORWARD PRECESSION (FORWARD=RED, BACKWARD=BLUE)



RAPP V2.60 Run: 10/21/2008 at 10:33:01, Plot File Name = PLOT.TXT

Mode Shape Plots (8) Plot # 2

Damped Eigenvalue Analysis

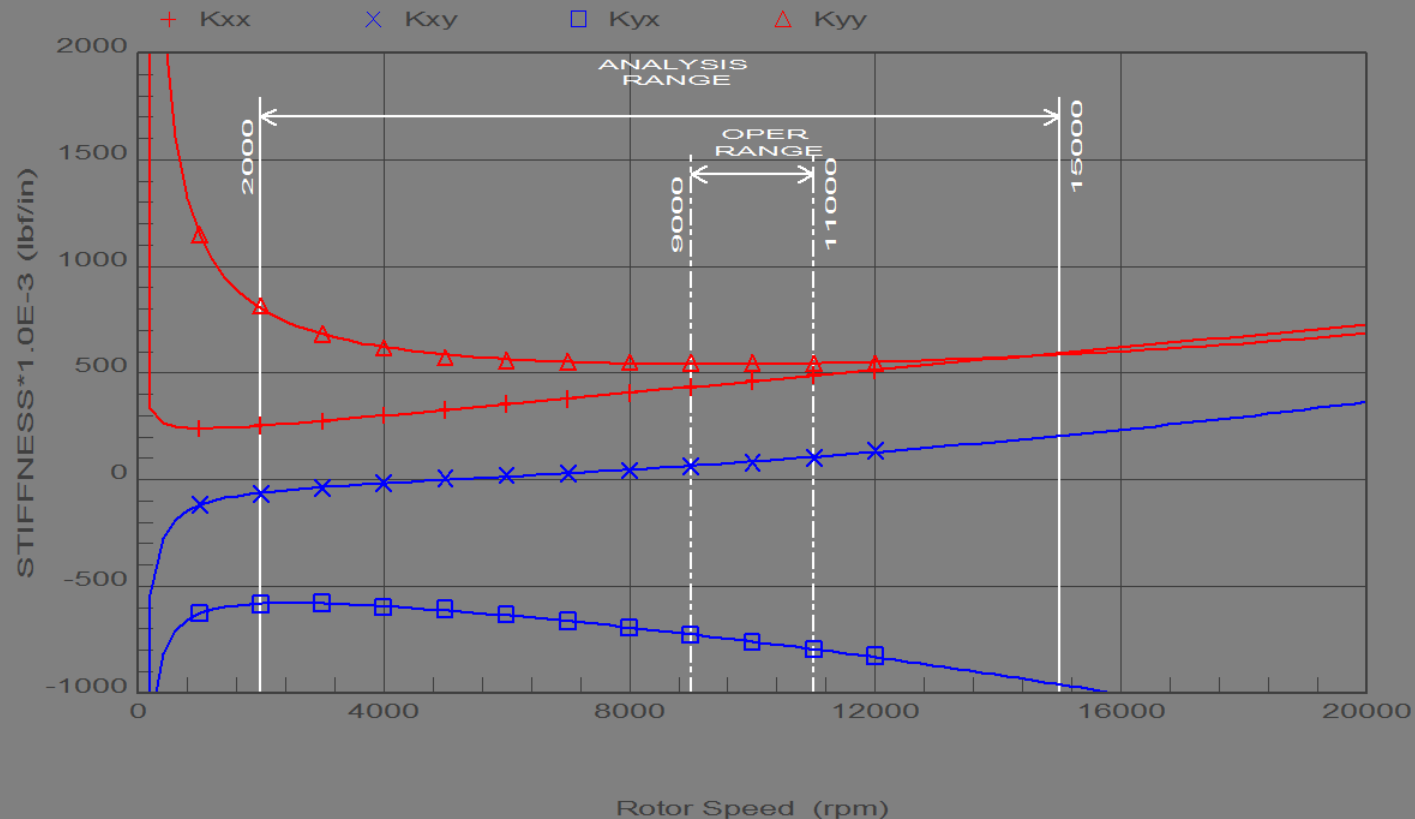
- **Required Information**
 - Structural Model
 - Transfer Functions
- **Analysis Assumptions**
 - No External Excitation
- **Analysis Results**
 - Natural Frequency Map
 - Stability Map
 - Mode Shapes (elliptical)
- **Why Perform Damped Eigenvalue Analysis?**
 - Provides Essential Frequency Survey to Locate Potential Synchronous and Non-Synchronous Critical Speeds
 - Provides only Steady State Assessment of Stability

Bearing Stiffness Values

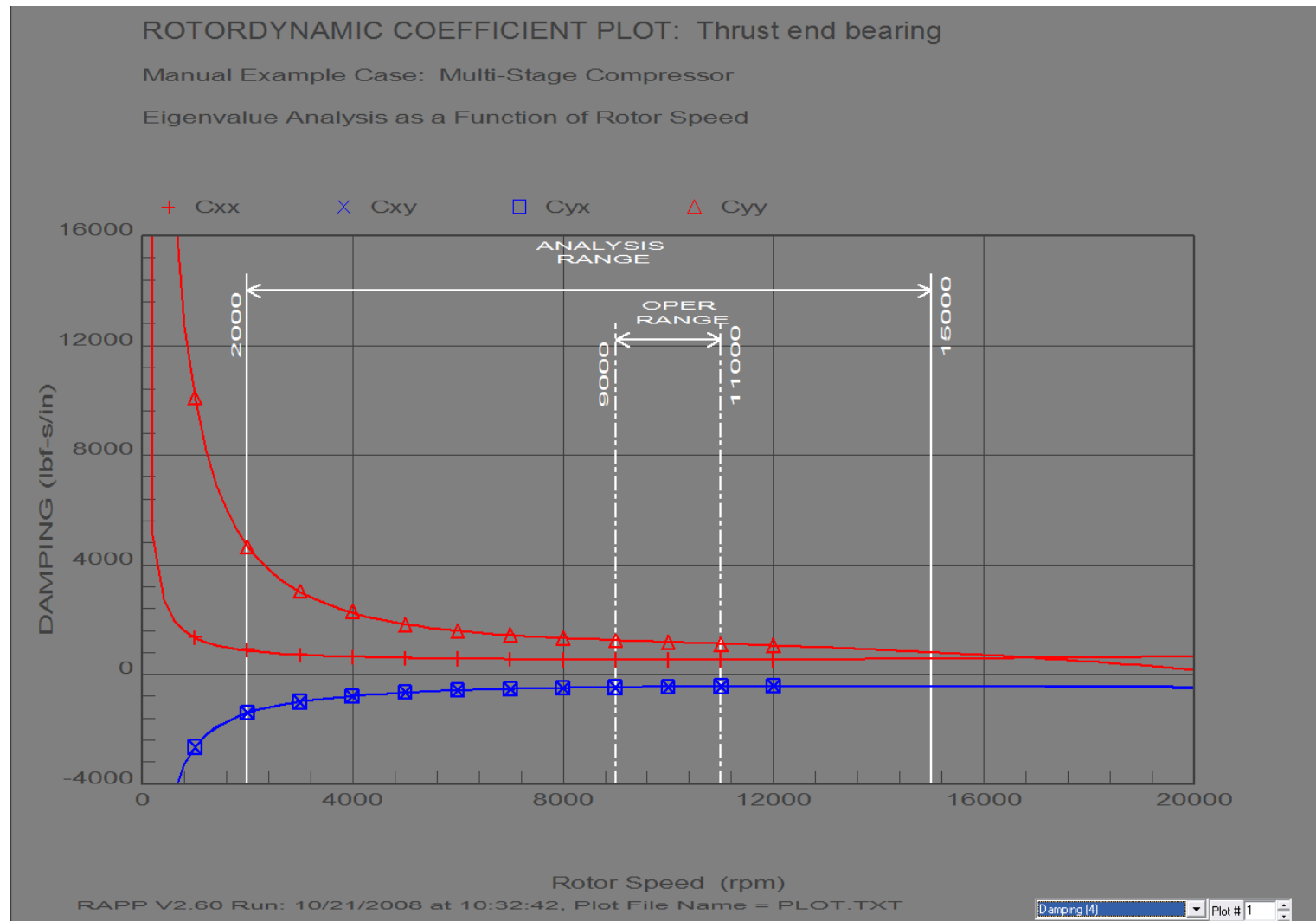
ROTORDYNAMIC COEFFICIENT PLOT: Thrust end bearing

Manual Example Case: Multi-Stage Compressor

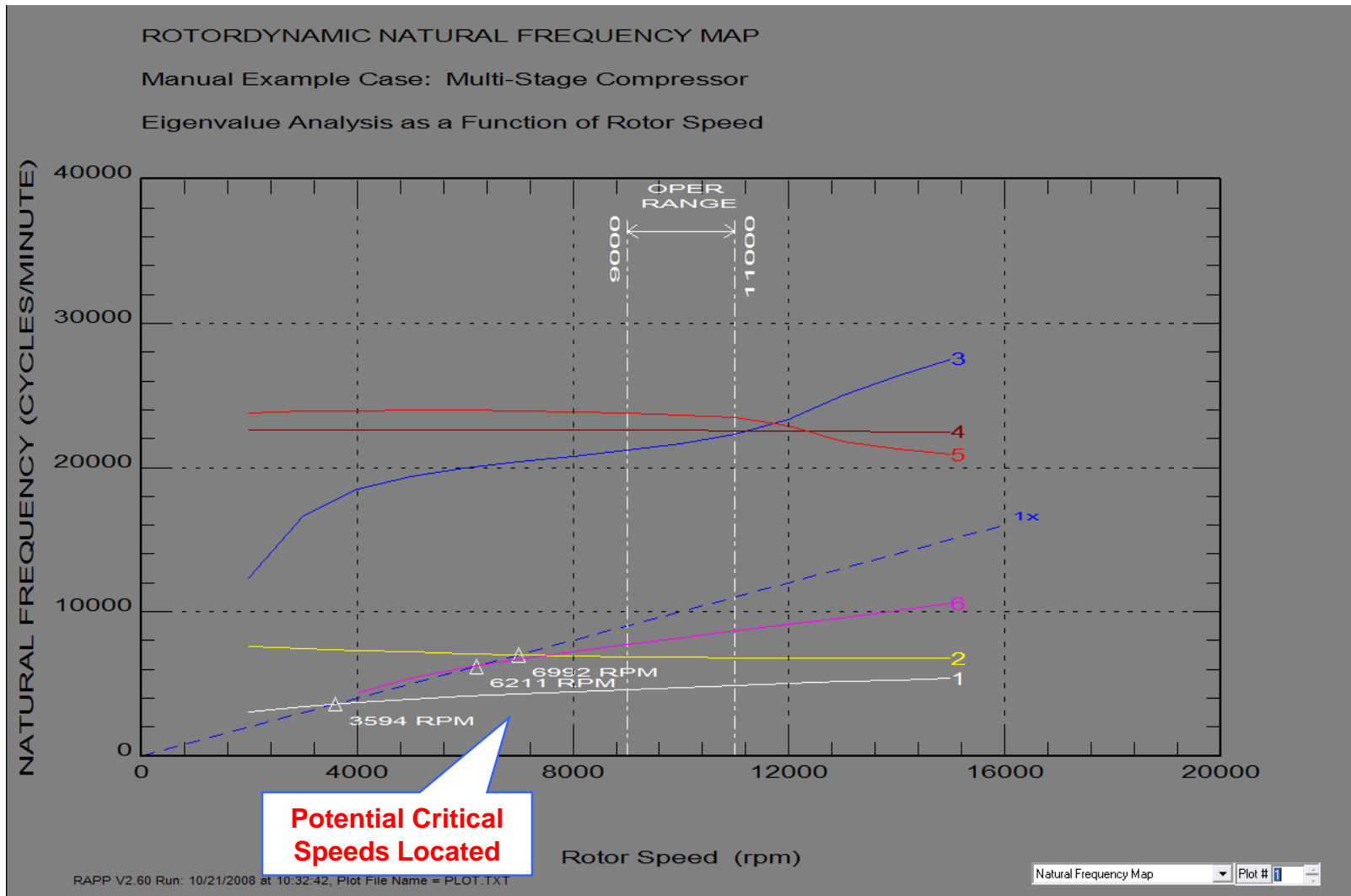
Eigenvalue Analysis as a Function of Rotor Speed



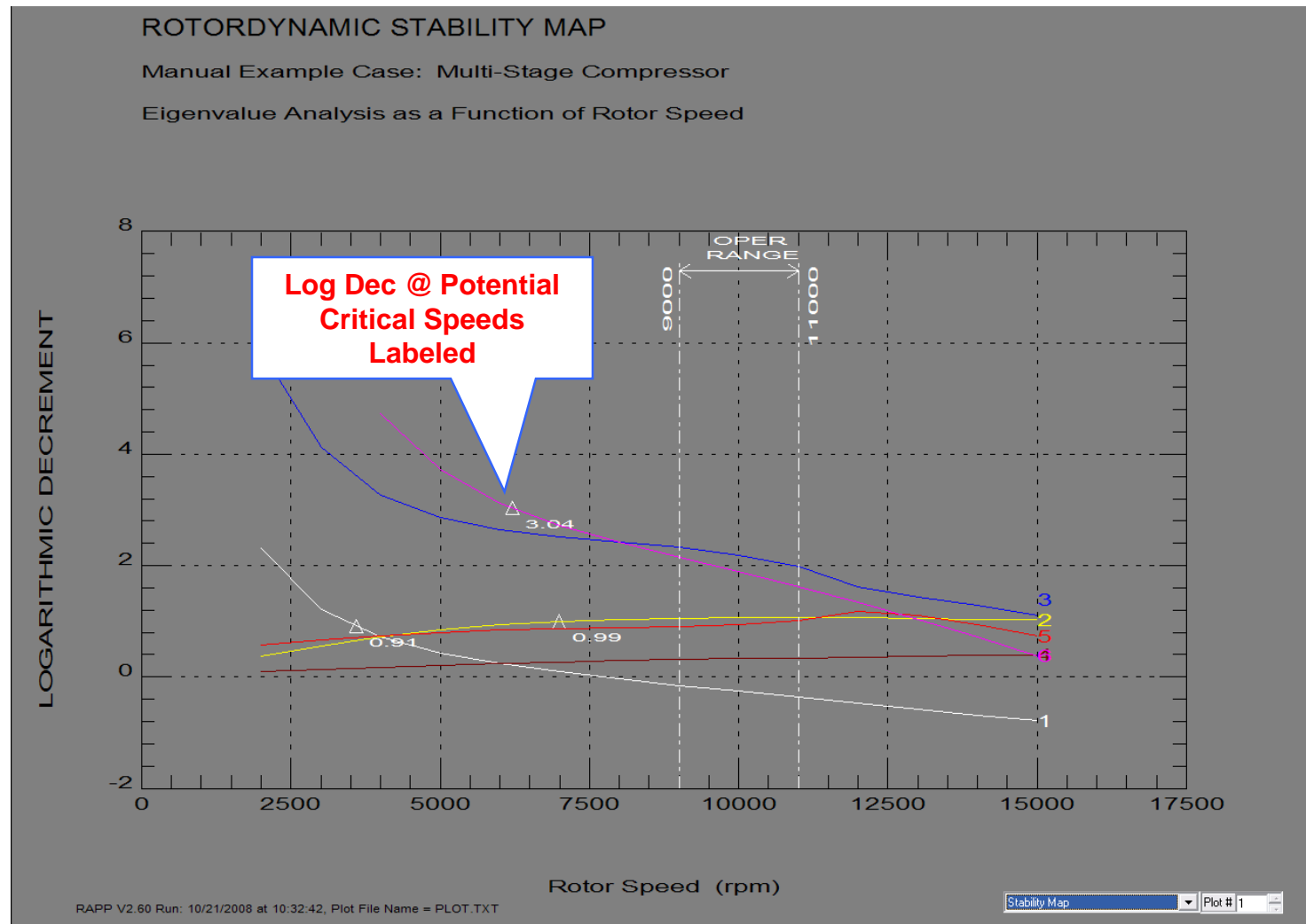
Bearing Damping Values



Natural Frequency Map



Stability Map



Mode Shape: Mode 1

ROTOR DYNAMIC MODE SHAPE PLOT - MODE #1

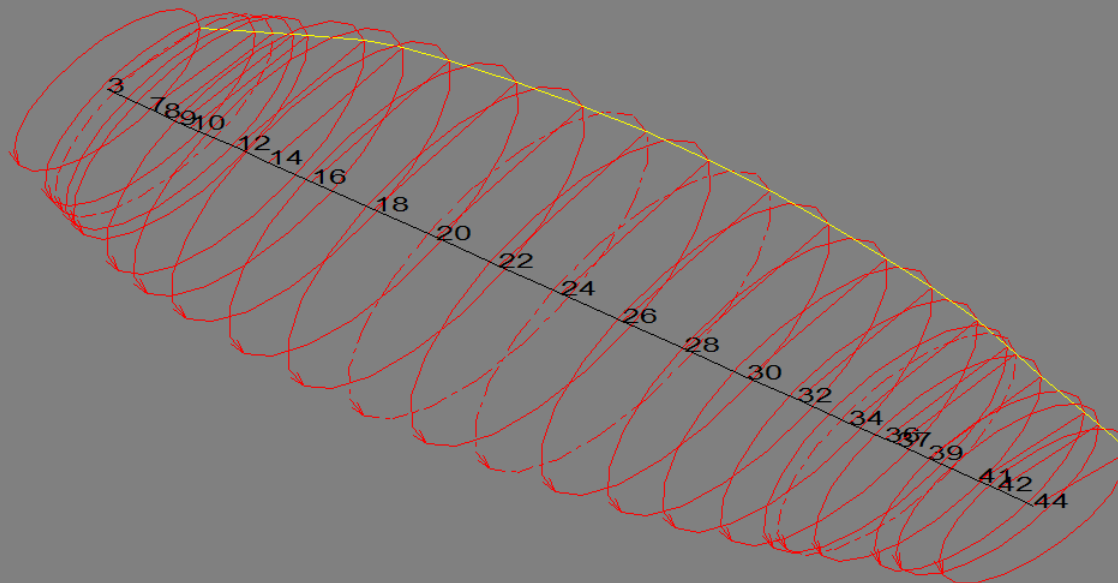
Manual Example Case: Multi-Stage Compressor

Eigenvalue Analysis as a Function of Rotor Speed

ANALYSIS POINT: Rotor Speed = 5000 (rpm)

NAT FREQ = 3930 cpm, LOG DEC = 0.429, POTENTIAL SYNC CRIT SPEED = 3594 rpm

STATION 22 ORBIT FORWARD PRECESSION (FORWARD=RED, BACKWARD=BLUE)



RAPP V2.60 Run: 10/21/2008 at 10:32:42, Plot File Name = PLOT.TXT

Mode Shapes (7) Plot # 1

Mode Shape: Mode 2

ROTOR DYNAMIC MODE SHAPE PLOT - MODE #2

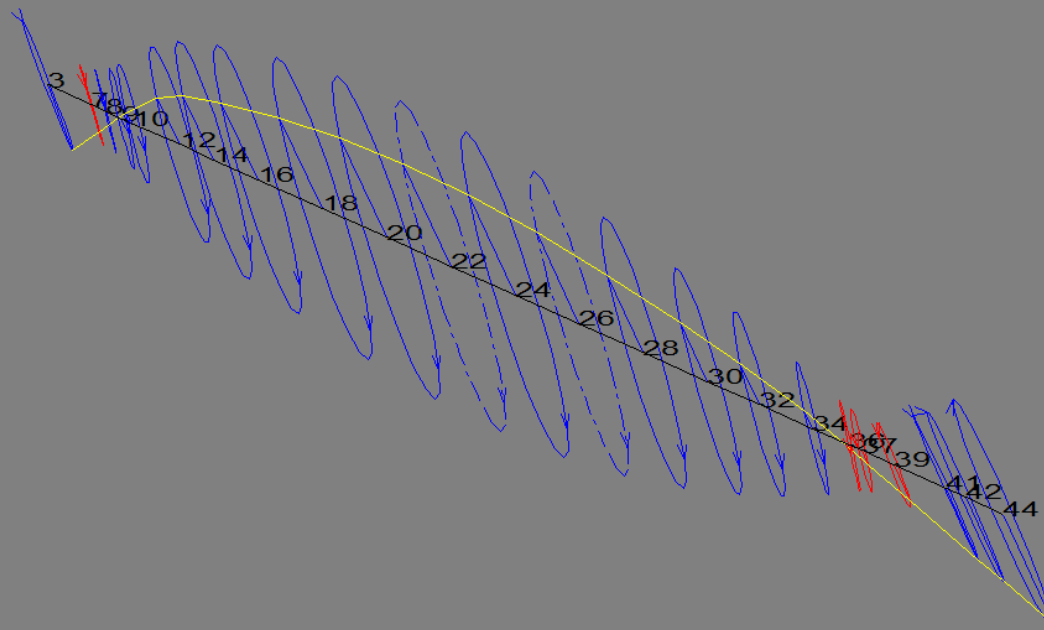
Manual Example Case: Multi-Stage Compressor

Eigenvalue Analysis as a Function of Rotor Speed

ANALYSIS POINT: Rotor Speed = 5000 (rpm)

NAT FREQ = 7192 cpm, LOG DEC = 0.845, POTENTIAL SYNC CRIT SPEED = 6992 rpm

STATION 22 ORBIT BACKWARD PRECESSION (FORWARD=RED, BACKWARD=BLUE)



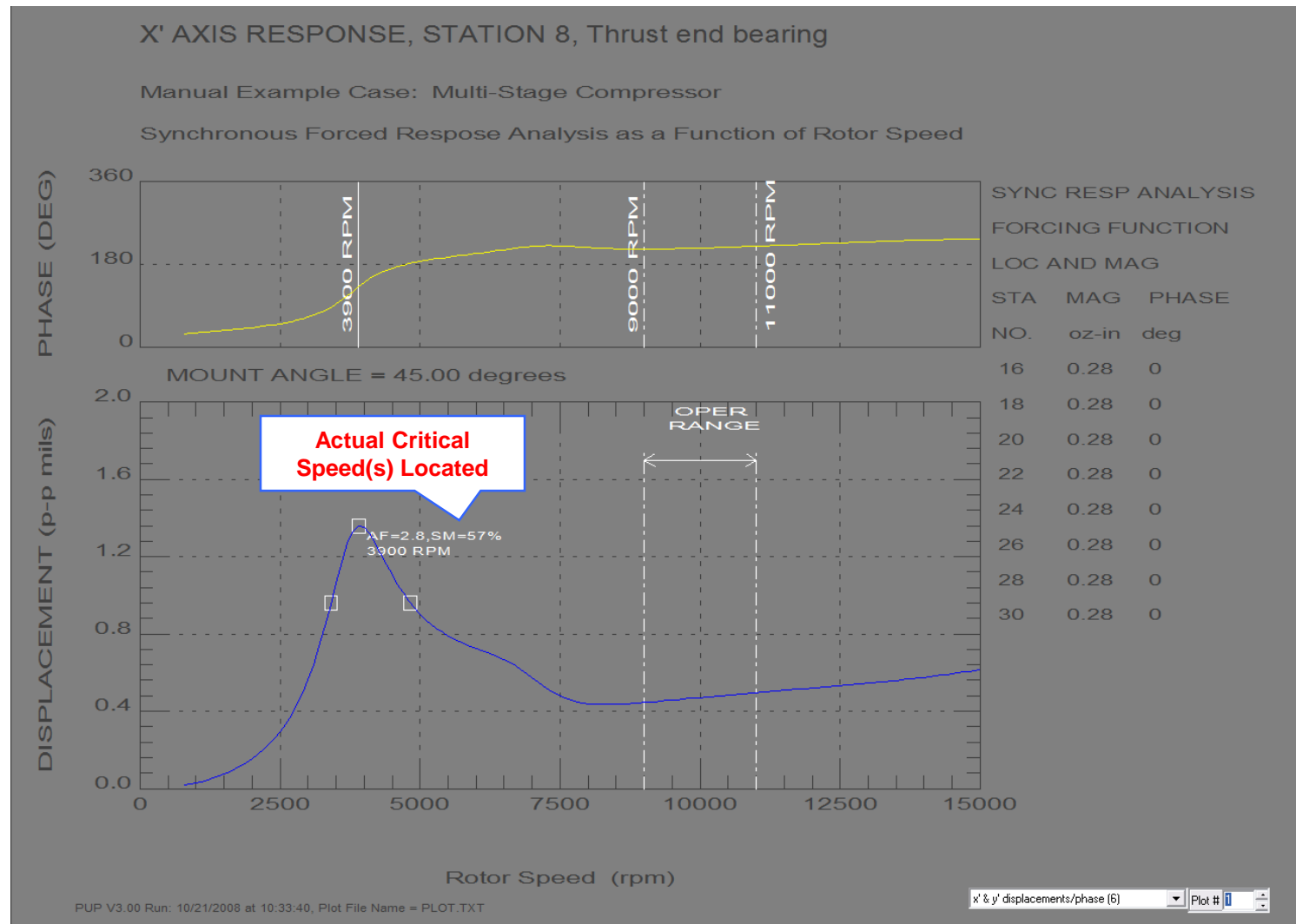
RAPP V2.60 Run: 10/21/2008 at 10:32:42, Plot File Name = PLOT.TXT

Mode Shapes (7) Plot # 3

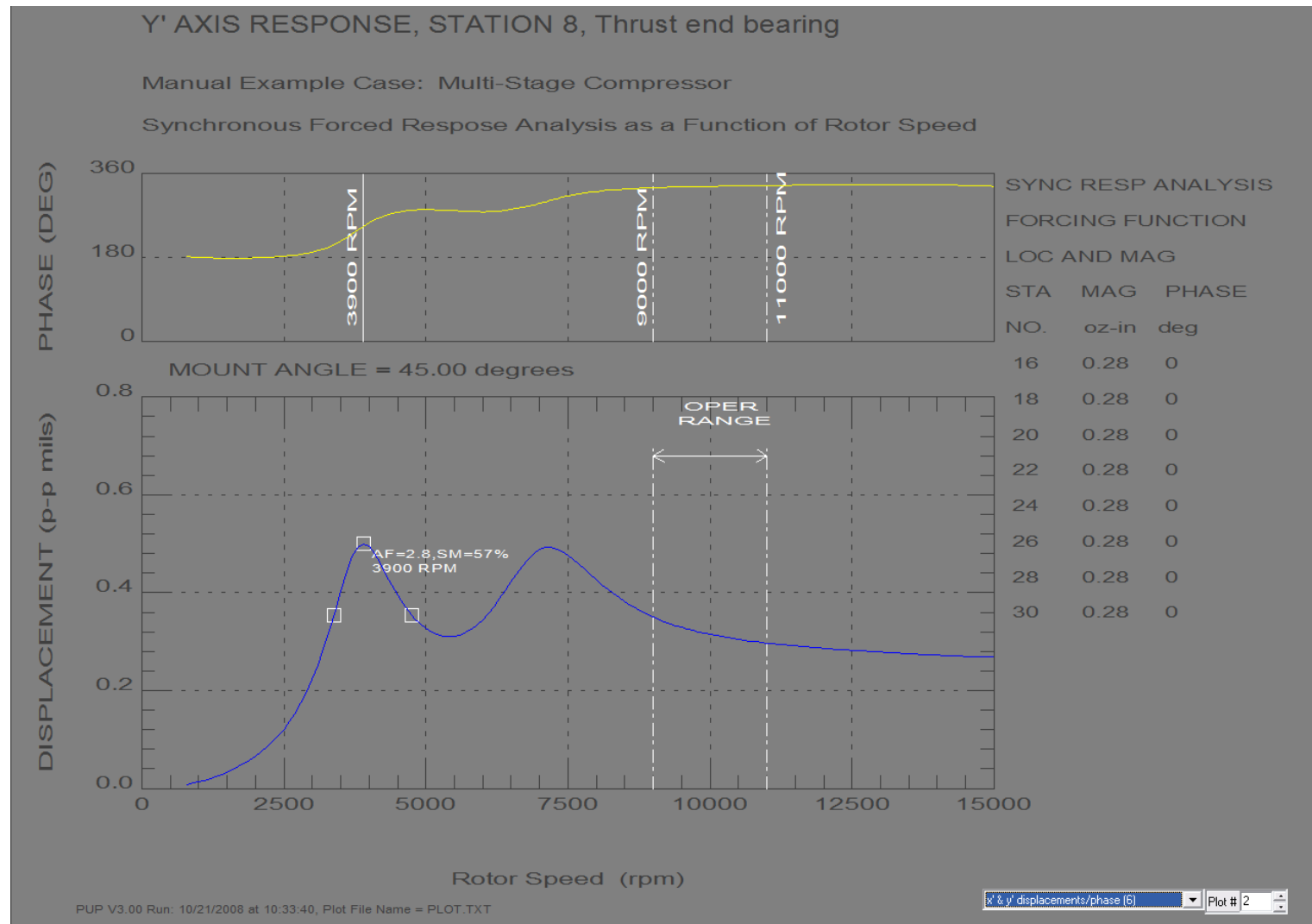
Forced Response – Steady State

- **Required Information**
 - Structural Model
 - Transfer Functions
 - Forcing Functions
- **Analysis Assumptions**
 - Unbalance Always Modeled
 - Other Forcing Functions Modeled as Needed
- **Analysis Results**
 - Vibration Amplitude
 - Dynamic Bearing Loads
 - Deflected Rotor Shapes (elliptical)
- **Why Perform Steady State Forced Response Analysis?**
 - Locate Actual Synchronous and Non-Synchronous Critical Speeds
 - Determine Amplification Factors
 - Establish Response Shapes

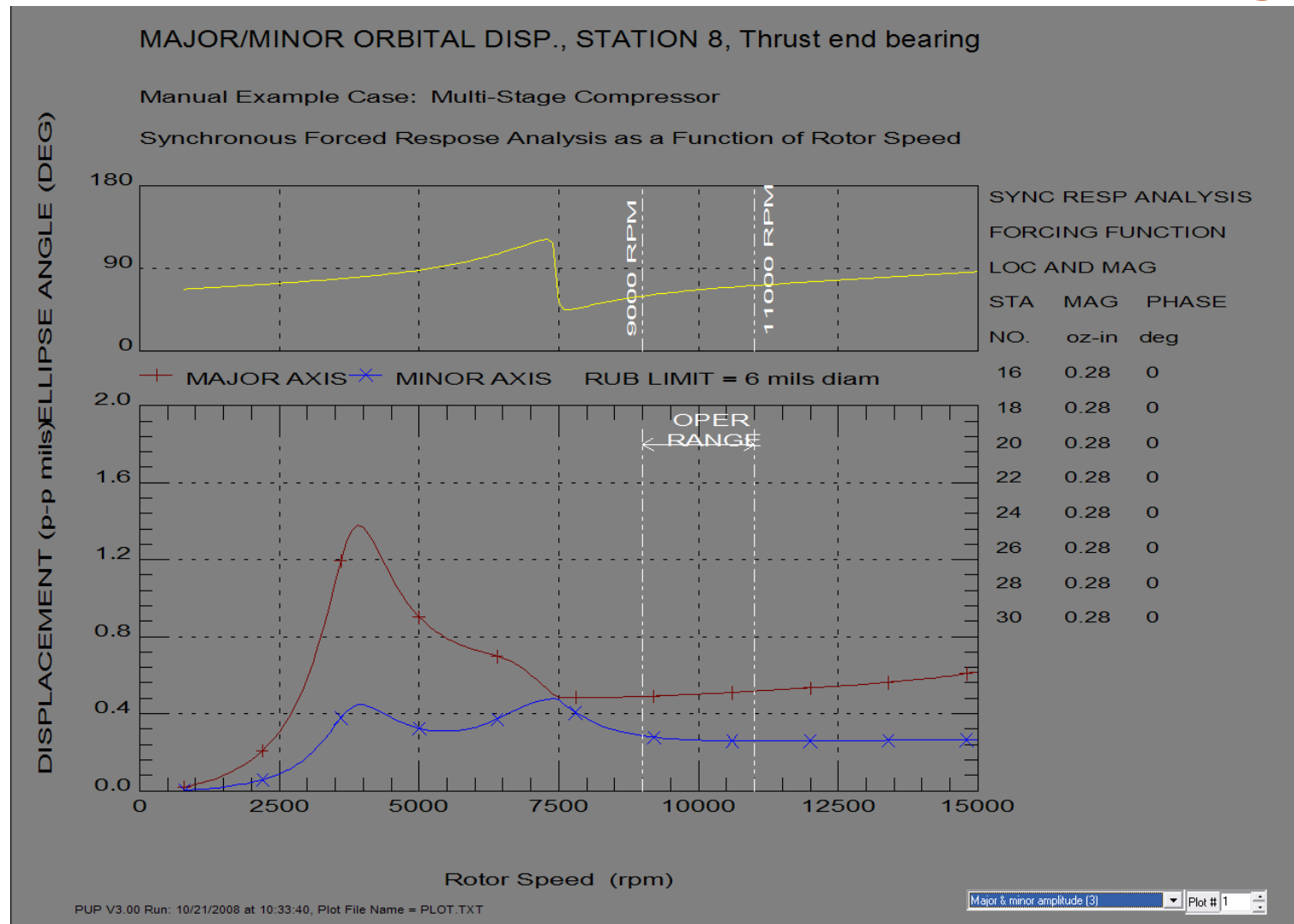
Horizontal Vibration @ Bearing



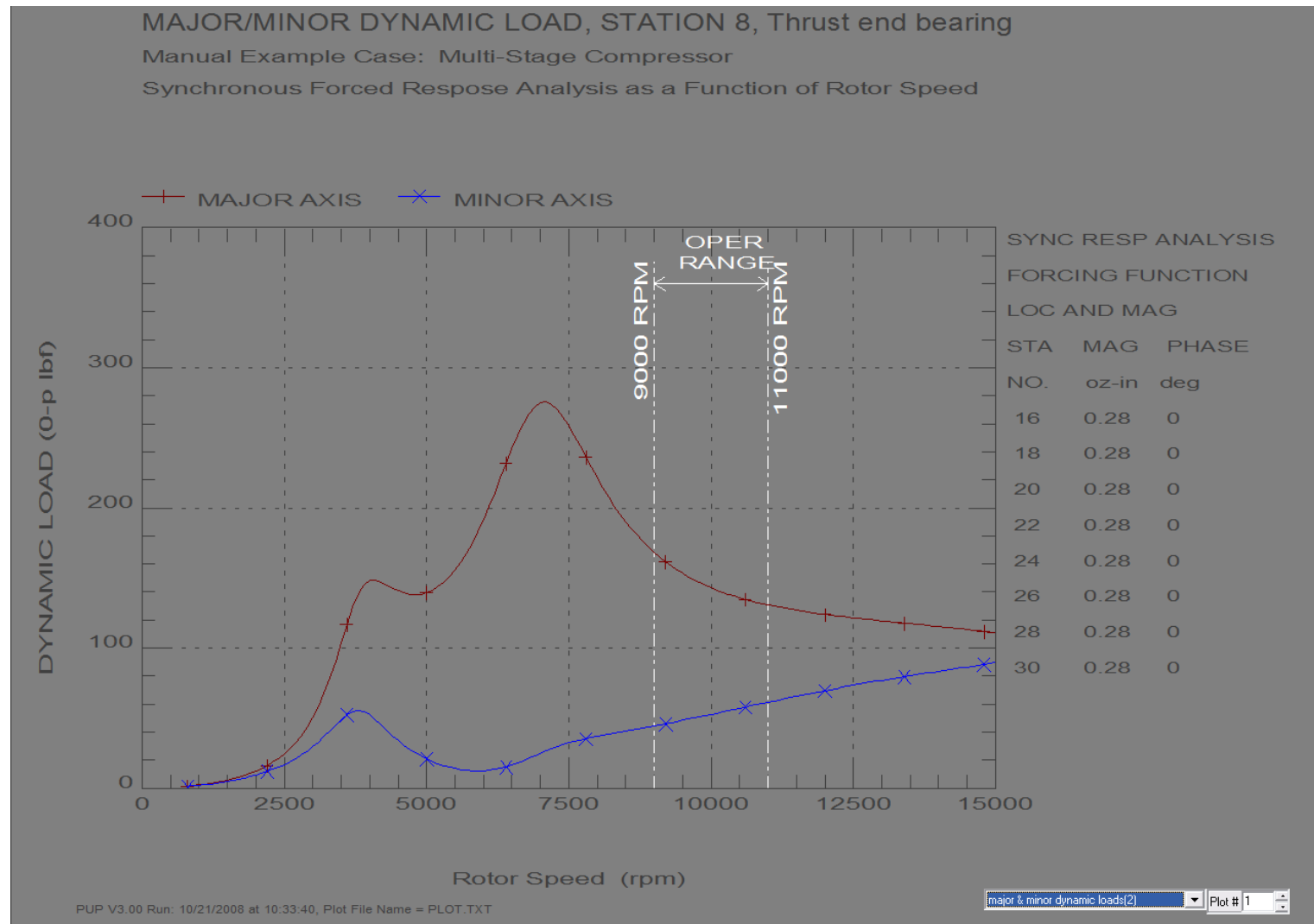
Vertical Vibration @ Bearing



Maximum Vibration @ Bearing



Maximum Dynamic Bearing Load



Rotor Response Shape

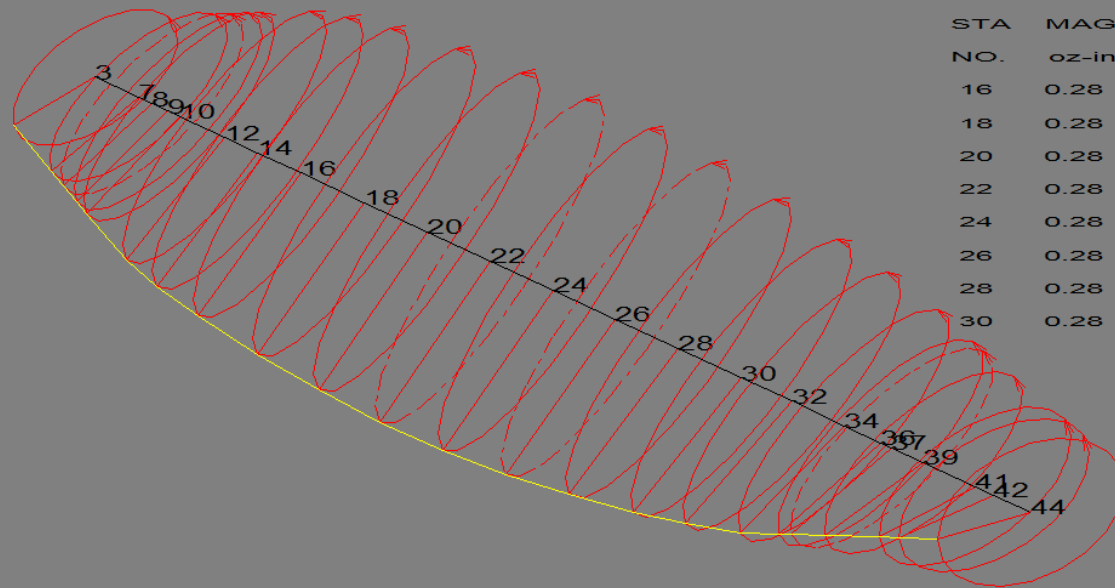
ROTOR RESPONSE SHAPE

ROTOR SPEED = 5000 rpm

Manual Example Case: Multi-Stage Compressor

Synchronous Forced Response Analysis as a Function of Rotor Speed

STATION 22 ORBIT FORWARD PRECESSION (FORWARD=RED, BACKWARD=BLUE)



SYNC RESP ANALYSIS

FORCING FUNCTION

LOC AND MAG

STA	MAG	PHASE
NO.	oz-in	deg
16	0.28	0
18	0.28	0
20	0.28	0
22	0.28	0
24	0.28	0
26	0.28	0
28	0.28	0
30	0.28	0

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3D lateral response shapes (2) Plot # 1